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AFOEHL REPORT 90-064EQ00094DEF



Source Emission Test of Gas Turbine Engine Test Facility Kelly AFB TX

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April 1990

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Final Report

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AF Occupational and Environmental Health Laboratory (AFSC)
Human Systems Division
Brooks Air Force Base, Texas 78235-5501

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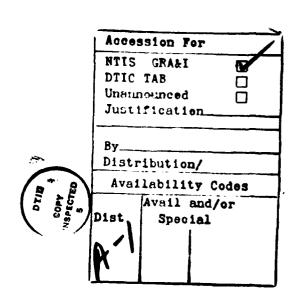
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I. INTRODUCTION

On 15 Nov 89, source emission testing for particulates, carbon monoxide (CO), total hydrocarbons (HC), sulfur dioxide (SO₂), and nitrogen oxides (NO_{χ}) was conducted on the Gas Turbine Engine Test Facility, Bldg 340, by personnel from the Air Quality Function of the AF Occupational and Environmental Health Laboratory Environmental Quality Division (AFOEHL/EQ). This survey was requested by SA-ALC/EM to determine emission data on various gas turbine engines. The emission data will be used to initiate a permit application for a new Gas Turbine Engine Test Facility. Personnel involved with on-site testing are listed in Appendix A.

II. DISCUSSION

A. Background

SA-ALC/EM requested emission data on various gas turbine engines. This information is required to initiate a permit application for a new Gas Turbine Engine Test Facility. No emission data could be located for two of the engines. SA-ALC/EM then requested source emission testing at the existing facility to determine emission data for these gas turbine engines.

B. Site Description

The Gas Turbine Engine Test Facility, Bldg 340, is a one story building located on Kelly AFB. The building is divided into several test cells (Fig 1). Cells 8 and 15A were selected for emission testing. The exhaust stacks for cells 8 and 15A are shown in Figures 2 and 3, respectively. Cell 8 was used to test gas turbine engine GTCP 85-180. Cell 15A was used to test jet fuel starter F-15 JFS.

C. Sampling Methods and Procedures

Federal Regulations require that stack emission testing be conducted in accordance with Appendix A and B to Title 40, Code of Federal Regulations, Part 60 (40 CFR 60). Determination of gas turbine engines emissions is to be done in accordance with Title 40, Code of Federal Regulations, Part 87. Therefore, sample train preparation, sampling and recovery, calculations and quality assurance were done in accordance with the methods and procedures outlined in 40 CFR 60 and 87.

Five sampling ports were installed on one side of the rectangular stacks resulting in five traverses of the stack cross-section. The ports on Cells 8 and 15A were installed approximately 1 and 1.5 duct diameters upstream and 2.8 and 4 duct diameters downstream from any flow disturbance, respectively. Based on the inside stack diameter, port location and type of sample (particulate), 25 traverse points (5 per traverse) were used to collect a representative sample.

Prior to the emission test on each stack, cyclonic flow was determined by using the Type S pitot tube and measuring the stack gas rotational angle at each traverse point. Flow conditions were considered acceptable when the arithmetic mean of the rotational angles was 20 degrees or less. A preliminary velocity pressure traverse was also accomplished concurrently.

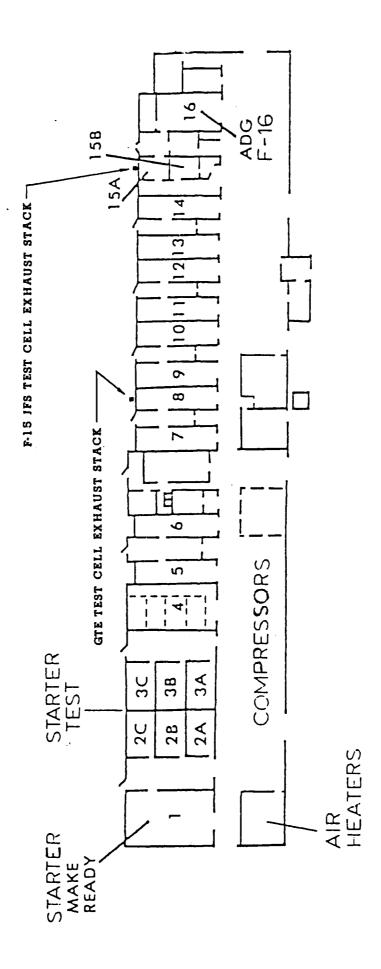


Figure 1. Schematic of Gas Turbine Engine Test Cell Facility



Figure 2. View of Cell 15A Exhaust Stack



Figure 3. View of Cell 8 Exhaust Stack

Particulates and sulfur oxides (SO_2) were collected and analyzed according to EPA method 8. Normally, three minimum one hour runs constitute a test; however, because jet fuel starters, as opposed to aircraft engines, can only operate for abbreviated periods, only one 62.5 minute run constituted a test.

The EPA method 8 sampling train (Figure 4) consisted of a button-hook probe nozzle, heated inconel probe, heated glass filter, impingers and a pumping and metering device. The nozzle was sized prior to each stack test so that the gas stream could be sampled isokinetically. In other words, the velocity at the nozzle tip was the same as the stack gas velocity at each point sampled. Flue gas velocity pressure was measured at the nozzle tip using a Type S pitot tube connected to a 10-inch inclined-vertical manometer. Type K thermocouples were used to measure flue gas and sampling train temperatures. The stainless steel probe liner was heated to minimize moisture condensation and the heated filter was used to collect particulates. The impinger train consisted of four impingers. The first and third impingers were standard Greenburg-Smith designs and the second and fourth impingers were modified Greenburg-Smiths. The first impinger contained an 80% solution of isopropanol and the second and third impinger each contained a 3% hydrogen peroxide solution. The fourth impinger contained silica gel. This system collected the water and SO2 from the gas stream. Moisture was subsequently determined according to EPA method 4. The pumping and metering system was used to control and monitor the sample gas flow rate.

Hydrocarbons were collected using a sampling train which consisted of a metal probe, two charcoal tubes connected in series, and a pump. The pump was calibrated before and after sampling.

Carbon monoxide (CO) was continuously monitored according to EPA method 10 using Neotronics (model CO101) direct reading equipment. However, ascarite and silica gel tubes were not used in the sampling train to remove CO_2 and water, since CO_2 and water were considered to be too low in concentration to be a significant interference in the CO infrared absorptance band used. Carbon monoxide monitor was calibrated using known concentrations of CO (0, 24, 55, and 100 parts per million). Calibration was accomplished according to EPA method 10. (Appendix F)

Nitrogen oxides were monitored using an Anarad Continuous Emission Monitor (CEM) according to EPA method 7E. The CEM consisted of a metal probe, heated umbilical, sample conditioner and NO_{X} analyzer. Nitrogen oxide analyzer was calibrated using known concentrations of NO (0, 50, 155, 250, and 348 parts per million). Calibration was accomplished according to EPA method 20. (Appendix F)

Oxygen (O_2) and carbon dioxide (CO_2) were collected and analyzed according to EPA method 3. The sampling train and ORSAT analyzer are shown in Figures 5 and 6.

EPA methods calculations were made using the Environmental Protection Agency publication entitled, "Source Test Calculation and check programs for Hewlett-Packard 41 Calculators," (EPA- $\frac{340}{1-85-013}$) and associated software programs. Equipment calibration data is shown in Appendix F.

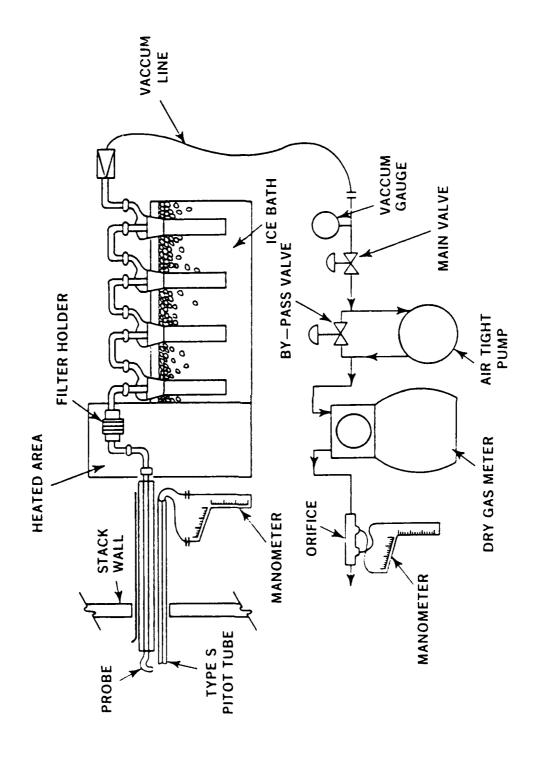


Figure 4. EPA Method 8/5 Sampling Train

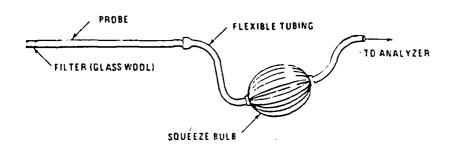


Figure 5. EPA Method 3 Sampling Train

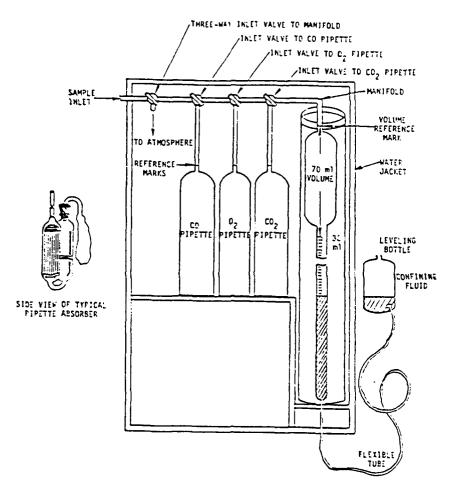


Figure 6. ORSAT GAS ANALYZER

D. Results

Table 1 provides a summary of the test results. Field data sheets are found in Appendix B and C and the resulting emissions data are presented in Appendix D. Tables 2 and 3 give engine operating parameters and emission rates. Emission rates were calculated according to 40 CFR 87 Subpart G, and these equations are shown in Appendix E.

TABLE 1. Summary of Test Results

Engine Tested	Vol, Part. Sample Part Vol (dscf)*	Part. (gdscf)+	CO ₂ (%)	O ₂ (%)	CO (ppm)	SO ₂ (mg/m³)	NO _x	HC (mg/m³)
F-15 JFS	72.26	1.49E-7	2.8	14.1	290.4	<.005	14.8	<.0067
GTCP85-180	49.06	6.45E-6	3.2	14.4	118.0	<.005	18.2	<.0067

^{*}dry standard cubic feet

TABLE 2. Emission Rates

Engine Tested	Power Setting (%)	Fuel Consumed (lbs/hr)	Parts. (1bs/hr)	CO (lbs/hr)	NO _x (1bs/hr)	HC (1bs/hr)
F-15 JFS	100	150	3.83E-6	3.08	0.258	<.0001
GTCP 85-18	0 100	360	2.48E-4	2.64	0.670	<.0001

Note: SO₂ was below detection limits

TABLE 3. Emissions Per Fuel Consumed

Engine Tested	Power Setting (%)	Parts. (1bs / 100	CO OO lbs fo	NO _x uel burn	HC ed)
F-15 JFS	100	2.55E-5	20.53	1.72	<.001
GTCP 85-180	100	6.89E-4	7.35	1.86	<.001

⁺grains per dry standard cubic feet

III. CONCLUSIONS

Both engines ran very clean; however, it is important to note that typically particulates and hydrocarbons concentrations are inversely proportional to the engine thrust or power setting; while nitrogen oxides and carbon monoxide emissions are proportional to thrust. Therefore, this test does not give an accurate indication of the maximum particulate and hydrocarbon emissions. However, maximum hydrocarbon emissions could be estimated by comparing the data of this report with emissions of other gas turbine engines.

IV. RECOMMENDATIONS

The Kelly AFB Gas Turbine Engine Test Facility, Bldg 340, is meeting Texas Air Control Board standards. No further action is necessary at this time.

The data contained in this report should be used as an estimate of emissions from the F-15 JFS and GTCP 85-180 and used to initiate a permit application for a new Gas Turbine Engine Test Facility.

REFERENCES

- 1. Standards of Performance for New Stationary Sources, Title 40, Part 60, Code of Federal Regulations, July 1, 1988.
- 2. Quality Assurance Handbook for Air Pollution Measurement Systems Volume III, Stationary Source Specific Methods, U.S. Environmental Protection Agency, EPA-600/4-77-027-b, Research Triangle Park, North Carolina, April 1977.
- 3. Source Test Calculations and Check Programs for Hewlett-Packard 41 Calculators, U.S. Environmental Protection Agency, EPA-340/1-85-018, Research Triangle Park, North Carolina, Sept 1985.
- 4. Control of Air Pollution From Aircraft and Aircraft Engines, Title 40, Part 87, Code of Federal Regulations, July 1, 1988.
- 5. Aircraft Engine Emissions, International Civil Aviation Organization, Annex 16, Vol II, June 1981.

APPENDIX A

Personnel

12

1. AFOEHL TEST TEAM

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Mr Dave Bartels SA-ALC/MATEF

AV 945-8711

COM (512) 925-8711

APPENDIX B
Field Data F-15 JFS

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AFOEHL/EQE

Date of report: Nov 15, 1989 11:46.04

Coverage of report: Nov 15,1989 10:10.00 to Nov 15, 1989 11:34.00

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APPENDIX C
Field Data GTCP 85-180

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	RUN NUMBER		DATE	15 1001 81	PLANT	" Kelly /	BILL	SAMPLE BON NUI	Aster T	METER BOX NUMBER	Qw/Qm		ೆ		TRAVERSE	NUMBER	_	7	3	, ,	5	1	9	8	0/	3	121	13	14	15	14	17	8)	19	3	166	4

				PART	ICIII ATE SA	TICILI ATE SAMPLING DATA SHEET	SHEET					
RUN NUMBER		SCHEMATIC OF STACK CROSS SECTION	OF STAC	K CROSS SI	CTION	EQUATIONS				AMBIEN	AMBIENT TEMP	
ሊ						0 = 0 F + 460	c					4o
DATE							,			STATIO	STATION PRESS	
15 Nov 89						_	5130-Fd-Cp.A 2					in Hg
PLANT							•	Ts. vp		HEATE	HEATER BOX TEMP	
Kell, AFR	7					.	1			_		4o
BASE										PROBE	PROBE HEATER SETTING	70
B12, 340	>											
SAMPLE BOX NOMBER										PROBE	PROBE LENGTH	
Nateul #2												in
9										NOZZL	NOZZLE AREA (A)	
												sq ft
Ow/Om										ථ	i	
ೆ										DRY GA	DRY GAS FRACTION (Fd)	9
TRAVERSE SAMPLING	\vdash	STATIC	STACK TEMP	EMP	VELOCITY	ORIFICE	GAS	GAS	GAS METER TEMP	dw:	SAMPLE	IMPINGER
		L	-	(Ts)	HEAD	DIFF.	SAMPLE	Z	AVG	PUO	жов	OUTLET
NUMBER (min)		(in H20)	(oF)	(oR)	(V _P)	PRESS.	VOLUME (cu ft)	(oF)	(T) (S) (S)	(o F.)	TEMP (OF)	TEMP (0F)
24 12.5	-	k	87 K		41.	2.44		76		85	273	66
7.5		-	269		6/,	2 44		7/6		70	272	79

																	1
$\frac{1}{1}$	-		-		-	-		-	-	-		-		<u> </u>	_	-	
\downarrow		Ц		_	_	_		_	L	_	_	_		_		L	
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و			. (1	.0.													
2,26			•														
144				1513													
				11	1												
167																	
4 =	\vdash						_								_	-	
+																	
-							_								_		
													_	_	4		18
												į					DEHL FORM 18
																	OEH

		EY DATA SHEET NO. 2 emperature Traverse)	
BASE Kelly		15 Nov 8	g
BOILER NUMBER		, , , , , , ,	
INCIDE STACK DIAMETER			Inches
INCIDE STACK DIAMETER 28.7 STATION PRESSURE 79.85			-a 11
STACK STATIC PRESSURE			
-, 05			In H20
TRAVERSE POINT NUMBER	VELOCITY HEAD, Vp IN H20	√ Vp	STACK TEMPERATURE (OF)
1	.13	0	394
2	.13	1	397
3	,!4	3	415
4	.16	2	420
5	, 20	4	420 426
	Ps = 28		
· · · · · · · · · · · · · · · · · · ·	T=410		
	3		
	Q.,	877	
	<u> </u>		
• • • • • • • • • • • • • • • • • • • •			
	·		
	AVERAGE		

	PRE	ELIMINARY SURVE (Stack G	Y DATA SH eometry)	IEET NO. 1
BASE		PLANT		
Kelly DATE 15 Nov 89	9	SAMPLING TEAM		
SOURCE TYPE AND MAK				
SOURCE NUMBER		equal 29	er 8-7	Inches
RELATED CAPACITY		Zipurt L	TYPE FUEL	Muite
DISTANCE FROM OUTSID	E OF NIPPLE TO INS	SIDE DIAMETER	<u> </u>	Inches
NUMBER OF TRAVERSES	5	NUMBER OF POINTS/TI	RAVERSE	(25 total Pouts
	Loc	CATION OF SAMPLING	POINTS ALON	
POINT	PERCENT OF DIAMETER	DISTANCE F INSIDE WA (Inches	LL	TOTAL DISTANCE FROM OUTSIDE OF NIPPLE TO SAMPLING POINT (Inches)
		3.6		
2		10.8		
3		18.0		
4		25.2		
5		32.4		
,				
7,				

	AIR POL	LUTI	ON PARTICUL	ATE ANA	LYTICAL	DATA			
BASE Ke	lly AFB	DATE	15 Nor	, 39		NI A			
BUILDING NUMBER	lly AFB GTE			SOURCE NI	GTE				
1.			PARTICU			IAL WEIGHT			
	ITEM		(gm)		(NII)	(gm)		(en)	
FILTER NUMBER			. 29.5	7	. 29	43		,0014	
ACETONE WASHIN Hell Filter)	GS (Probe, Front	·	98,75	549	98,	7358		- 0191	
BACK HALF (II nee	oded)							•	
			Total We	ight of Partic	culates Colle	ected		,0205 "	
И			WATE		r				
	ITEM		FINAL WE		INIT	AL WEIGHT		WEIGHT WATER (gm)	
IMPINGER 1 (H20)			72		20	00		-128	
IMPINGER 2 (H20)			230		7.0	SO		-128 30	
IMPINGER 3 (Dry)			241		200			41	
IMPINGER 4 (Silica	Gel)		233		2	٥٢)	33		
_			Total Wei	ight of Water	Collected			e m	
111.	1	T	GASES						
ITEM	ANALYSIS 1		ANALYSIS 2	ANAL	YSIS 3	ANALYSIS 4		AVERAGE	
VOL % CO2	3.1		3.2	3	.2			3.2	
VOL % 0 ₂	14.3		14.4	14	1.4			3.Z 14,4	
VOL % CO									
VOL % N ₂									
		Vol %	N ₂ = (100% - % C		% CO)				

AFOEHL/EQE

Date of report: Nov 15, 1989 15:52.45

Coverage of report: Nov 15,1989 13:50.00 to Nov 15, 1989 14:54.00

	Gas		Avg	Max	Min	n
Stream			_			
Stack	NOX	PPM	18.2	21.1	13.5	64
Stack	CO	PPM	134.4	139.0	115.0	64
Corrected	to Ca	libratio	n Plots			
Stack	NOX	PPM	18.2	21.1	13.5	64
Stack	CO	PPM	118.0	123.0	96.0	64

APPENDIX D

Emission Data

EPA Method 5

	RUN #1	RUN #2
Meter Box Y	.9990	•9990
Delta H (in H_{20}	5.12	2.26
Bar Press (in Hg)	29.85	29.85
Meter Vol (FT ³)	73.9670	50.6250
MTR TEMP F	86	86
Static HOH IN	04	 05
ml Wate	59.5	76
% нон	3.7	6.8
% CO ₂	2.8	3.2
% Oxygen	11	14.4
MWd	29.20	29.09
MW Wet	28.78	28.33
SQRT PSTS	10.8739	12.7063
Time Min	62.5	62.5
Nozzle Dia	.500	•377
STK Dia IN	25.0	28.7
VOL MTR STD	72.261	49.064
Stack DSCFM	2,990	4,479
% Isokinetics	96.72	101.64

EPA Mass Flow

	RUN 1 F-15JFS	RUN 2 GTCP 85-180
VOL MTR STD	72.261	49.064
Stack DSCFM	2,990	4,479
Front 1/2 mg	.0007	.0205
GR/DSCF	1.49E -7	6.45E -6
mg/MMM	3.42E -4	0.01
lb/HR	3.83E -6	2.48E -4
KG/HR	1.74E -6	1.12E -4

LABORATOR	Y ANALYSIS REPORT	AND RECOR	O (General)	DATE 0 5	DEC 1989
TO:		FROM			
SAMPLE IDENTITY				20	NOV 89
SAMPLE FROM				i	NTROL NR
BLDG 340				1 65 33	74 - 388
PD- 680					
				CESULTS	
OEHLF	Bnst #	Volume Sumptoul	ning FRANT SECTION	my BACIC SLAIN	דפדזוג יוין/מיני
65384	S x 8 900 8 3	7.38L	N O	NO	CM
65385	5 X8 900 8'4	7.38L	NO	No	NO
6538 w	SX8400 85	1.50L	MO	Wo	NO
65387	5X8900 86	7.50L	NO	Mo	NO
65388	BK840087		NO	Mo	
ND=NONE DE	EFECTED		LIMIT OF	DETECTION _ C	0.005 mg
ANDREW RICHAF Chief, IH Analysis	EDSON III CC 14				
REQUESTING AGENCY (Mailing A	ddreso)				
OEHL /EGE					
ATT CAPT VAUGIL	Ni l				

AF3C FORM 3511, DEC 85 REPLACES AMD FORM 641, SEP 82, WHICH IS OBSOLETE

APPENDIX E
Emission Equations

CALCULATION OF MASS EMISSION RATES

1. Mass emission rates for gas turbine and piston engines were calculated using the formulas specified by EPA.

For gas turbine engines, these formulas are:

HC emission rate =
$$\frac{M_{HC} \frac{(HC)}{10^4} F}{(M_C + \alpha M_H) \left[\frac{(CO)}{10^4} + (CO_2) + \frac{(HC)}{10^4}\right]}$$

CO emission rate =
$$\frac{M_{CO} \frac{(CO)}{10^4} F}{(M_C + \alpha M_H) \left[\frac{(CO)}{10^4} + (CO_2) + \frac{(HC)}{10^4} \right]}$$

$$NO_{x} \text{ emission rate} = \frac{\frac{M_{NO_{2}} \frac{(NO_{x})}{10^{4}} F}{(M_{C} + \alpha M_{H}) \left[\frac{(CO)}{10^{4}} + (CO_{2}) + \frac{(HC)}{10^{4}} \right]}$$

* Defined - next proje

2. Emission rates for hydrocarbons, carbon monoxide, oxides of nitrogen, and particulates also are expressed as pounds per 1000 pounds of fuel. These values are obtained by dividing the emission rate, expressed as pounds per hour, by the fuel flow (pounds per hour) and multiplying by 1000.

where:

CO emission rate = Pounds per hour of exhaust carbon monoxide emitted in an operational mode

 ${
m NO}_{\mathbf{X}}$ emission rate = Pounds per hour of exhaust oxides of nitrogen emitted in an operational mode

 M_{HC} = Molecular weight of methane, M_{HC} = 16.04

 M_{CO} = Molecular weight of carbon monoxide

 M_{NO_2} = Molecular weight of nitrogen dioxide

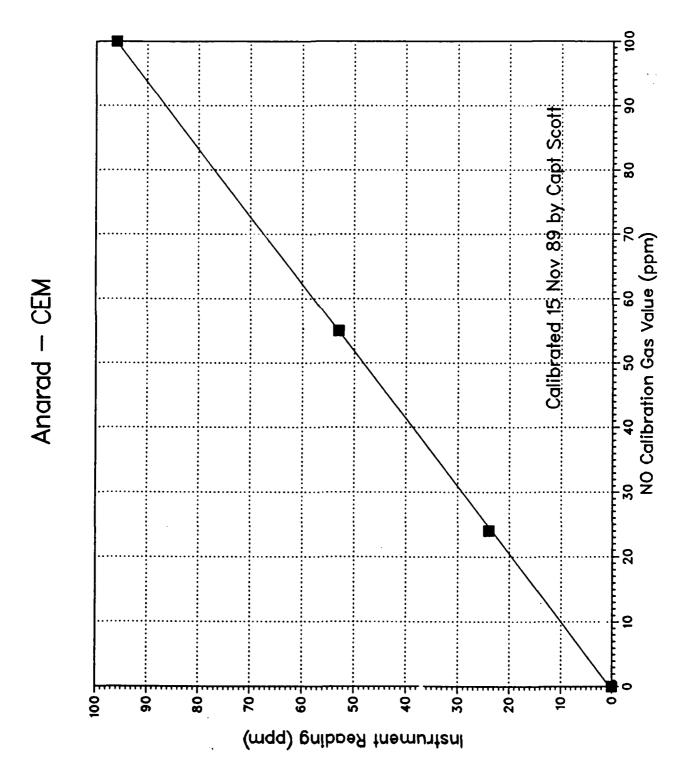
 M_C = Atomic weight of carbon

 $M_{\rm H}$ = Atomic weight of hydrogen, $M_{\rm H}$ = 1.008

- α = Atomic hydrogen carbon ratio of fuel (equal to 2 in approximation equations)
- (HC) = Concentration of hydrocarbons in the exhaust sample in parts per million carbon equivalent, i.e., equivalent propane × 3.
- (CO) = Concentration of carbon monoxide in the exhaust sample in parts per million by volume.
- (CO₂) = Concentration of carbon dioxide in the exhaust sample in volume percent
- (NO_X) = Concentration of oxides of nitrogen in the exhaust sample in parts per million by volume, $NO + NO_X$.
 - F = Mass rate of fuel flow in pounds per hour.

APPENDIX F
Calibration Data

Calibrated 13 Nov 89 by S\$gt Schillings Neotronics - Model # CO101 150 175 200 225 250 275 300 CO Calibration Gas Value (ppm) 75-450-350-Instrument Reading (ppm)



POSTTEST DRY GAS METER CALIBRATION DATA FORM (English units) Sec # / Vary hy

Plant Pre Kelly Fl. Pod Kack GTE.
Pretest Y 6,999 MO: 1.89 Date 12 Dec 39 Meter box number Nutech 2 Dry gas meter number 1/4 Barometric pressure, $P_b = 29.350$ in. Hg

Y	۲	$V_{\mathbf{q}} P_{\mathbf{b}} (t_{\mathbf{d}} + 460)$	$V_d \left(P_b + \frac{\Delta H}{13.6}\right) \left(t_w + 460\right)$	(10X29.35X531.25) (10,174X3354.0368(525)	3,	(10) (24)	$\mathbf{v} = (0.992)$	
		>		0.93	0.92	2669	Υ =	
		Vacuum	setting, in. Hg	4.0	4.0	4.0		
			(0), min	25,51	25.52	15.57		
	Dry gas meter	ry gas meter	Inlet Outlet Average (t.). (t.).	4°	57.12	0.47	78.0	
ure			: Outlet). (t.).	, do Ч	0.01 611 00.01	61166	80 805 75 755	
emperature		Inlet (t.).	, d	211 26	2015 2013	300 80.5 X 1		
Ţ	Wet test	meter (t).	, A G	50 69	5.40 PO	67 67		
Gas volume	Dry gas	meter (V.)	ft ₃	HL1: 01	10.225	10.277		
	Wet test	meter (V),	ft ₃	10	10	9		
Orifice	manometer	setting, (△H).	in. H ₂ 0	.5.	5,	,5		

If there is only one thermometer on the dry gas meter, record the temperature under $\mathsf{t}_{\mathbf{d}}$

 $V_{\rm w}$ = Gas volume passing through the wet test meter, ft³.

 $V_d = Gas$ volume passing through the dry gas meter, ft³.

 $t_{\rm w}$ = Temperature of the gas in the wet test meter, oF.

= Temperature of the inlet gas of the dry gas meter, OF.

= Temperature of the outlet gas of the dry gas meter, oF.

 t_d^{\prime} = Average temperature of the gas in the dry gas meter, obtained by the average of $t_{d_1^{\prime}}$ and $t_{d_2^{\prime}}$, °F.

 $\Delta H = Pressure differential across orifice, in. <math>H_2^{0}$.

 Y_1 = Ratio of accuracy of wet test meter to dry gas meter for each run. Y = Average ratio of accuracy of wet test meter to dry gas meter for all three runs; tolerance = pretest Y ±0.05Y. 999±,049% $\Rightarrow B_194$ $\leftarrow U_1 \Rightarrow U_104$ 401/ LaRy 176:01

 P_b = Barometric pressure, in. Hg.

 θ = Time of calibration run, min.

Quality Assurance Handbook M4-2.4A

Exercit

METER BOX CALIBRATION DATA AND CALCULATION FORM

(English units)

Date 28 Supt 89 Meter box number Nulech 2

Barometric pressure, Pb = 29.83 in. Hg Calibrated by Sectl & Vaugha

		Gas v	olume	Temperature		[emperature				
	Orifice	Wet test	Dry gas	Wet test		gas met				
	manometer	meter	meter	meter	Inlet	Outlet	Avg	Time		
	setting	(V _w),	$(v_{d}),$	(t _w),	(t _{d.}),	(t _d),	(t _d),	(Θ),	<u></u>	••••
IAC	(ΔΗ), in. Η ₂ Ο	ft ³	ft ³	°7′ R	"FR	97 K	SFR.	min	Yi	ΔH_0 in. H_2^i 0
J	0.5	5	5.06¢	78 538	79 84 54 1,5	77 79 53 8	537.8	12.9	0.990	1.897
Ll	1.0	5	5.06\$	79 539	87 9,549	80 81 540.5	<i>5 44.</i> B	9.\$	0.996	1.837 1.840
L	1.5	10	10.150	96 19 539.5	96 98 557	86 875465			1.004	1,943
᠘	2.0	10	10.195	l 7ú	418	07	553,5		1.00 £	1,744
L	3.0	10	10.155	79 80 539,5	101 547.5	90 9/ 550 .5	556.5	10,7	1.008	1,710
7	4.0	10	10, 623		80 89544,5	74 77535.5	540	10.0	0,791	a, 383
								Avg	0,799	1.969

ΔH, in. H ₂ O	<u>ΔΗ</u> 13.6	Y _i =	$\frac{V_{w} P_{b}(t_{d} + 460)}{(P_{b} + \frac{\Delta H}{13.6}) (t_{w} + 460)}$	$\Delta H_{i}^{0} = \frac{0.0317 \Delta H}{P_{b} (t_{d} + 460)} \left[\frac{(t_{w} + 460) \Theta}{v_{w}} \right]^{2}$
0.5	0.0368		5) (39.83)(539.8) 5.06) (39.83+0°)(3.6)(538)	Hay = (0.0317)(.5) [(538) (12.9)]
1.0	0.0737	<u> </u>	5) (29.82) (544.8) (06) (29.82) 13.6) (539)	Ha= (.03,7) (1) (5.39.6) (9.0)]2
1.5	0.110	Ц	(s) (29.82) (551.8) (3.15) (29.82+ 1.5/3.6)(539.5)	Ha, = (-03:7)(1.5) [(539.5)(15.2)]2
2.0	0.147	(4	10) (29.82) (553.5) 10.195) (29.82 + 3.86)(539)	Hey= (29.82) (553.5) (53.2) 10
3.0	0.221	<u> </u>	(10) (29.8.2) (556.5) (0.155) (29.82+7,36)(539.5)	Has= (29.82) (556.5 (10.7) 72
4.0	0.294	6	10.072) (34.87) (240) 10.072) (34.87 + 1/37)(2382	(.0317) (40) (15345) (10.0) 72

 $^{^{\}mathbf{a}}$ If there is only one thermometer on the dry gas meter, record the temperature under $\mathbf{t}_{\mathbf{d}}$.

Quality Assurance Handbook M4-2.3A (front side)

CALIBRATION DATA FOR PUMPS A and B

	<u>BEFORE</u>	AFTER
PUMP A	123 cc/min	124 cc/min
PUMP B	123 cc/min	127 cc/min

Pump A used in determining total hydrocarbons on F-15 JFS.

Pump B used in determining total hydrocarbons on GTE.

cc/min = cubic centimeter per minute

Distribution List

	Copies
HQ AFLC/SGBE Wright-Patterson AFB OH 45433-5001	1
HQ USAF/SGPA Bolling AFB DC 20332-6188	1
HQ AFSC/SGP Andrews AFB DC 20334-5000	1
7100 CSW Med Cen/SGB APO New York 09220-5300	1
OL AD, AFOEHL APO San Francisco 96274-5000	1
USAFSAM/TSK/ED/EDH/EDZ Brooks AFB TX 78235-5301	1 ea
Defense Technical Information Center (1) Cameron Station Alexandria VA 22304-6145	2
HQ HSD/XA Brooks AFB TX 78235-5000	1
HQ USAF/LEEV Bolling AFB DC 20330-5000	1
HQ AFESC/RDV Tyndall AFB FL 32403-6001	1
SA-ALC/EM Kelly AFB TX 78241-5000	5